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PART V. THE SMALL SCALE OF TH	F PHOTOGRAPH AND THE LARGE	; ·
AMOUNT OF HALATION SEVEPELY PE	STRICT THE VALIDITY OF	
PHOTO INTERPRETATION RESULTS.	DETAILS SUCH AS ENGINE	OF CYS
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AS TO THE AIRCRAFT'S ROLE AND CHARACTERISTICS ARE LARGELY SURMISES BUT ARE PRESENTED HEREIN FOR CONSIDERATION. THE LARGE, STRAIGHT WING WITH AN AREA ON THE PART VI. ORDER OF 2500 SQ. FT. AND NOMINALLY, THOUGH NOT EXCEED-INGLY, HIGH ASPECT RATIO OF ABOUT 7.5 CAN APPLY TO SEVERAL DIFFERENT DESIGN CONSIDERATIONS. IN ONE CASE, SUCH A WING WOULD BE DESIRABLE FOR A SLOW FLYING AIRCRAFT DESIGNED TO LIFT A VERY HEAVY LOAD OFF THE GROUND WITH REASONABLE TAKEOFF DISTANCES, CRUISE AT ONLY NOMINAL ALTITUDES AT SLOW SPEEDS, AND LAND WITH A HEAVY LOAD. SECOND APPLICATION WOULD BE FOR AN AIRCRAFT TO FLY AT HIGH ALTITUDES AT SUBSONIC SPEEDS. A VERY LOW WING LOADING AND HIGH ASPECT RATIO WOULD BE THE MOST DESIRABLE COMBINATION FOR SUCH A VEHICLE. STRUCTURAL PROBLEMS AND A DESIRE FOR SLIGHTLY GREATER SPEEDS COULD DICTATE A COMPROMISE, RESULTING IN ONLY A MODERATELY HIGH ASPECT RATIO FOR SUCH AN AIRCRAFT.

PART VII. CONSIDERING THE FIRST CONCEPT, THAT OF A HEXVY VEHICLE, THE OBVIOUS IMPLICATION IS THAT OF A FIRST GENERATION, NUCLEAR POWERED AIRCRAFT. A CONFIGURATION OF THIS TYPE WOULD PRESENT THE MINIMUM PROPULSION SYSTEM REQUIREMENTS FOR A GIVEN WEIGHT. IT IS PROBABLE THAT THE VEHICLE WOULD BE FLIGHT TESTED FOR SOME TIME, USING CHEMICAL FUEL, PRIOR TO INTEGRATION OF THE NUCLEAR

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PART IX. THE POSSIBILITY THAT THE AIRCRAFT COULD BE NOTHING MORE THAN A BULL, CAMP, CAT, CUB OR COOT HAS BEEN CONSIDERED. PHOTO ANALYSIS RESULTS COMBINED WITH EXAMINATION OF TRANSPORT ACTIVITY MAKE SUCH A POSSIBILITY RATHER REMOTE.

PART X. FURTHER ANALYSIS OF THE CONFIGURATION OBTAINED

FROM PRESENT PHOTOGRAPHY WOULD BE RATHER

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